# Research Article

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# Exergy Analysis of Oxy-Biogas Combustion with Different Rates of the CO<sub>2</sub> Content

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## **Abstract**

This work presents the exergetic evaluations of the oxy-biogas combustion process in an adiabatic furnace for the cases of biogas involving different carbon dioxide (CO<sub>2</sub>) concentrations. A three-dimensional steady-state computational fluid dynamics model was developed for combustion simulation. The model was first verified with the data of oxy-natural gas and showed good agreement with the data. Thus, simulation studies were performed for the oxy-biogas combustion with different CO<sub>2</sub> concentrations of biogas fuel, from 10 vol% to 40 vol% with a 10% increment, at a constant input power capacity. The results show that the sum of specific thermo-mechanical and chemical exergy of combustion products has a decreasing trend with increasing CO<sub>2</sub> content in biogas. However, the exergy flow rates of the combustion products increased with the increase in CO<sub>2</sub> due to the increasing mass flow rates. Increasing the CO<sub>2</sub> level in biogas led to an increase in the chemical exergy fraction of the combustion products. Thus, the exergy loss fractions resulting from incomplete combustion varied increasingly from 5.8 % to 13.8 % in the range from 10 % CO<sub>2</sub> to 40 % CO<sub>2</sub> contents of biogas.

**Keywords:** Biogas; oxy-combustion; chemical exergy; exergy destruction; computational fluid dynamics.

## 1. Introduction

The growing global need for energy, the depletion of fossil fuels, and environmental damage have led to increased interest in the use of renewable energy sources. Among green energy sources, biogas is an energy source that can be produced from a wide variety of organic wastes such as municipal waste, agricultural and industrial waste, and sewage sludge. In response to the search for inexhaustible, sustainable, and clean energy, which is one of today's important problems, biogas is a strong candidate because it can be used as a renewable electricity and heat source when needed, stored, and produced from organic waste. Biogas, consisting of combustible methane-CH<sub>4</sub> (50-75%), noncombustible carbon dioxide-CO<sub>2</sub> (25-50%) and small amounts of other gases, and water vapour, is a valuable energy source with a capacity of 5.5-7 kWh m<sup>-3</sup> [1, 2]. Methane becomes a harmful greenhouse gas when released into the atmosphere, but it directly affects the amount of energy in biogas and is also an important component for reducing harmful emissions and fossil fuel use. For example, it is known that biogas, when used as vehicle fuel, emits less carbon monoxide, hydrocarbon, and nitrogen oxide emissions than gasoline or diesel engines [3].

According to research, the design of the biogas production process and the type of substrate used determine the composition of the gas. Depending on the type of raw material, methane yield varies between 51% and 65%. For example, the methane efficiency of grass silage is 54%, cattle

manure is 60%, liquid pig manure is 65%, forage beet is 51%, and corn silage is 52%. However, depending on the feedstock, biogas may contain contaminants including halogenated compounds, sulphur compounds, nitrogen, and organic silicon species [4]. As can be understood, the high and pure methane content of biogas is an important parameter in energy conversion by burning. The methane content determines the calorific value of biogas. The average calorific value of biogas is quite low, at about 21.5 MJ/m³, compared to the calorific value of natural gas, which is 36 MJ/m³. By removing non-flammable CO<sub>2</sub> from the biogas composition, higher calorific values of biogas can be obtained when the amount of methane increases [5].

Some cleaning technologies, such as cryogenic process, water cleaning, and membrane, can be used to improve biogas and increase its efficiency. However, gradual cleanup and upgrade processes increase costs. Another solution to improve the biogas combustion process is to reduce CO<sub>2</sub> concentration by blending biogas with different fuels [6, 7]. It is possible to come across examples of blending various fuels and biogas used for this purpose in the literature. Methane, oxygen, propane, hydrogen, microalgae biofuels supported by nanoparticles, and biodiesel are examples of these fuels [8-11].

Another strategy is an oxygen-enriched or oxy-fuel combustion process to improve biogas combustion. Oxygen-enriched biogas combustion has been investigated in the literature, although it is not at a sufficient level. The studies

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performed by Yılmaz et al. [12] and Safe et al. [13] show that the stability limits of biogas are improved, and flame temperature increases with oxygen enrichment. In addition to oxygen-enriched and oxy-fuel combustion of biogas, blending of biogas with different fuels under oxygenenriched and oxy-fuel combustion studies are also available in the literature, such as the use of diesel-biogas blend with oxygen-enriched in an engine [14], biogas/kerosene mixture with oxygen enrichment in a mini gas turbine engine [15], and biogas-propane-oxygen enrichment [16]. In these studies, it was concluded that oxygen enrichment had a significant effect on the entire reaction process and that enrichment of air with O2 caused an increase in the laminar combustion rates of both fuels examined, respectively. As a summary of the studies in the literature, it was determined that hydrogen addition and oxygen enrichment have positive effects in reducing combustion instability. In biogas combustion studies using hydrogen and oxygen, it was found that CO emissions decreased, and NOx emissions increased with O<sub>2</sub> enrichment, the net heat release rate of the biogas flame increased with the increase in the net reaction rate of the biogas, and more stable combustion was achieved [17-19].

However, the evaluation of these strategies is a critical issue to determine which strategy is more useful and its improvement potential for sustainable energy solutions in such combustion systems. In this regard, exergy flow analyses, sometimes referred to as the second law analysis, become crucial in the evaluation of systems [20, 21]. Exergy analysis is a useful tool for increasing a system's overall efficiency because it can indicate the origins, extent, and locations of thermodynamic inefficiencies [22]. Therefore, researchers have pointed out that the second law of thermodynamics should be used to make a better evaluation of the systems they examine to reveal the destruction of exergy in the systems [23].

Terhan and Comakli [24] performed the exergy analysis of a natural gas-fuelled boiler used for district heating. They reported that the maximum exergy destruction occurred in the combustion chamber, and the exergy efficiency of the boiler was 32.77% while energy efficiency was 82.21%. Costa et al. [25] carried out an exergy analysis of a thermal power plant with a 50 MW capacity using residual forest biomass fuel. They stated that the main irreversibility occurred in the combustion chamber, and the potential for enhancing the system efficiency is by reducing the moisture content in the biomass. Zueco et al. [26] conducted the exergy analysis of a power plant using different fuels, noctane, methane, propane, and soybean-based biodiesel. Similarly, they concluded that the main source of the exergy destruction stems from the combustion process in the boiler. Besides, it is stated that the fuel with a simple molecular structure and oxygen will reduce the exergy destruction, although exergy efficiencies were reported at about 31% for all fuels used in the study.

Along with the exergy analyses for boiler and power plant combustion systems, much research is also available on the exergy analyses on internal combustion engines for different combustion strategies and fuels [27-29]. Li et al. [30] studied the exergy analysis of three different combustion modes in internal combustion engines, namely conventional diesel combustion (CDC), homogeneous charge compression ignition (HCCI), and reactivity

controlled compression ignition (RCCI). Wang et al. [31] performed an extensive study on a turbocharged hydrogen engine to map exergy efficiencies for varying engine speed and load. They reported that the hydrogen-fuelled engine is limited theoretically by 59% exergy efficiency. In general, many of these studies reported that most of the exergy loss occurs during the combustion process.

Literature survey shows that there are numerous studies performed exergy analyses of systems involving the combustion process. However, the exergy analysis of oxybiogas combustion in a chamber is missing in the literature, which is a research gap in this field. Besides, no studies have evaluated the effect of CO2 content on oxy-biogas combustion in terms of exergy. The current study aimed, therefore, to investigate the exergy analysis of oxy-biogas combustion in a furnace designed for natural gas system oxycombustion presented in the study of Yin et al. [32], to reveal the effect of CO<sub>2</sub> content of biogas during oxy-fuel combustion considering both thermo-mechanical and chemical exergy, to determine the exergy loss due to incomplete combustion by varying CO2 content. Thus, the current study contributes to the literature in this field by presenting findings based on exergy analyses of oxy-biogas combustion, which is a proposed strategy to overcome drawbacks, such as the lower calorific value of biogas and combustion stability encountered during the combustion

#### 2. Materials and Methods

In the present work, combustion modeling of biogases consisting of different rates of  $CO_2$  content was numerically carried out under oxy-fuel conditions in a furnace. The combustion model was simulated using the ANSYS-Fluent CFD software tool, which employs the control volume method. Thus, the combustion characteristics and exergetic investigations were performed by the simulated results. The detailed model development and exergy analysis procedure are given in the following sections.

# 2.1 Geometric Model and Model Formulation

A three-dimensional furnace model was created to simulate the oxy-combustion by considering the dimensions of a 0.8 MW oxy-natural gas flame furnace in the reference [32]. The furnace has a horizontal design, a square cross-section with 1.05 m x 1.05 m, and a 3.44 m length. The furnace has a burner consisting of a fuel inlet with a 16 mm diameter and two annulus having 28mm and 36mm outer diameters for air entrance. An exhaust line with a 500 mm diameter and a 740 mm length was also designed for the exhaust gas exit. The furnace was modeled in its quarter segment to reduce the computational time since the model can be defined by symmetry boundaries. Thus, the model was meshed with the hexahedral elements for the computational domain as shown in Figure 1.

The model formulations consist of conservation equations of mass, momentum, energy, and transport equations in three-dimensional cartesian coordinates. The equations were solved in the computational domain in which flow with chemical reactions takes place, to estimate the flow field, thermal, and combustion products. The conservation equations and transport equations are expressed for steady-state and Newtonian fluids as follows [33]:

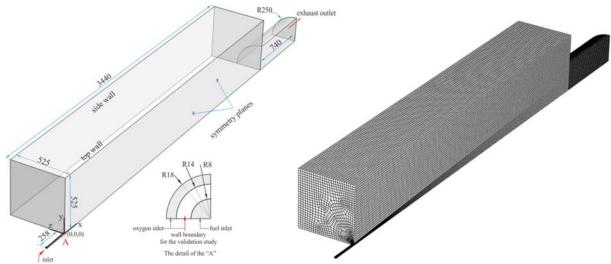


Figure 1. The modelled furnace and its grid structure.

Continuity equation

$$\nabla(\rho\vec{V}) = 0 \tag{1}$$

Momentum equation

$$\nabla (\rho \vec{V} \vec{V}) = -\nabla p + \nabla \cdot (\tau) + \rho \vec{g} + \vec{F}$$
 (2)

where,  $\vec{V}$  is the velocity vector,  $\rho$  is the density of the fluid. In Eq. (4), p and  $\tau$  represent the static pressure and stress tensor, respectively.  $\vec{F}$  denotes the external body forces, and  $\rho \vec{g}$  is the gravitational force.

Energy equation

$$\nabla \cdot \left[ \vec{V} (\rho E + p) \right] = \nabla \cdot \left[ k_{eff} \nabla T - \sum_i h_i \vec{J}_i + \left( \tau_{eff} \cdot \vec{\vartheta} \right) \right] + S \quad (3)$$

where, T is temperature and  $k_{eff}$  is defined as the effective conductivity.  $\vec{J_i}$  is the diffusion flux of species i, E is total energy, h is sensible enthalpy, and S is the energy source due to the combustion reaction. The total energy is defined as:

$$E = h - \frac{p}{\rho} + \frac{v^2}{2} \tag{4}$$

The enthalpy for ideal gas mixture is expressed as:

$$h = \sum x_i h_i \tag{5}$$

In Eq. (5),  $x_i$  is the mass fraction of i th species in a mixture.

The mass fraction of the species in combustion reactions is determined by the species transport equations, and the conservation equation becomes in the following form [34]:

$$\nabla \cdot (\rho \vec{V} x_i) = -\nabla \cdot \vec{J}_i + R_i \tag{6}$$

where  $R_i$  and  $\vec{J_t}$  are the net production rate and the diffusion flux of i th species in chemical reaction,  $\vec{J_t}$  is defined as follows:

$$\vec{j}_i = -\left(\rho D_{i,m} + \frac{\mu_t}{Sc_t}\right) \nabla x_i \tag{7}$$

where  $D_{i,m}$  represents the diffusion coefficient for species i, and  $Sc_t$  is the turbulent Schmidt number, which is set to 0.7 [35].

The present work employs the Eddy-Dissipation-Concept (EDC) model, incorporating chemical kinetic mechanisms, to determine the term  $R_i$  in Eq. (6) [36]. In the EDC model, chemical reaction occurs in small turbulent scales defined by the fine scales; the length fraction of the fine scales and the volume fraction of the fine scales. They are calculated as follows [37]:

$$\gamma^* = C_{\gamma} \left(\frac{\nu \epsilon}{\nu^2}\right)^{1/4} \tag{8}$$

where, \* indicates the quantities of the fine scales. v,  $\epsilon$ , and k are the kinematic viscosity, the dissipation rate, and turbulent kinetic energy, respectively.  $C_{\gamma}$  is a time scale constant. The volume fraction of the scales is estimated as  $\gamma^{*3}$ . A variety of species react in the fine structures across the time scale calculated by Eq. (9).

$$\tau^* = C_{\tau} \left(\frac{\nu}{\epsilon}\right)^{1/2} \tag{9}$$

The  $C_{\tau}$  is a time scale constant. The constants,  $C_{\gamma}$  and  $C_{\tau}$  were set to 1.0 and 3.0 [38] for the current study.

The present work used the modified 4-step global combustion mechanism of Jones and Lindstedt [32, 39]. The mechanisms with Arrhenius equation parameters, which determine the related reaction rate, are given in Table 1. The Arrhenius expression is given by Eq. (10).

$$k_r = A T^{\beta} \exp\left(-\frac{E}{RT}\right) \tag{10}$$

In Eq. (10),  $k_r$  represents the reaction rate constant of reaction r. A and  $\beta$  are the pre-exponential factor and temperature exponent, respectively. E and R denote the activation energy for the relevant reaction and the universal gas constant, respectively

The flow field and temperature distribution in the combustion environment are essential to well predict the species since reactions occurring interact with turbulent and heat transfer mechanisms [40]. Therefore, the turbulence and radiation effects were considered by the models; the standard k- $\epsilon$  turbulence model and the P1 radiation model in the present study.

Table 1. The modified global mechanism of Jones and Lindstedt (JL\_modified).

No	Reactions	Rate exponent	A	β	E [j/kmol]
R1	$CH_4 + 0.5O_2 \rightarrow CO + 2H_2$	$[CH_4]^{0.5} [O_2]^{1.25}$	4.4 1011	0	1.26 10 <sup>8</sup>
R2	$CH_4 + H_2O \rightarrow CO + 3H_2$	$[CH_4][H_2O]$	$3.0\ 10^8$	0	$1.26\ 10^8$
R3	$H_2 + 0.5O_2 \leftrightarrow H_2O$	$[H_2] [O_2]^{0.5}$	5.69 1011	0	$1.465 \ 10^8$
R4	$CO + H_2O \leftrightarrow CO_2 + H_2$	[CO] [H <sub>2</sub> O]	$2.75\ 10^9$	0	$8.36\ 10^{8}$

The standard k-  $\epsilon$  model has two transport equations for the turbulence kinetic energy (k) and the dissipation rate ( $\epsilon$ ), which are expressed in Eq. (11) and (12) [34, 41].

$$\frac{\partial}{\partial x_i} (\rho k V_i) = \frac{\partial}{\partial x_i} \left[ \left( \mu + \frac{\mu_t}{\sigma_k} \right) \frac{\partial k}{\partial x_i} \right] + G_k + G_b - \rho \epsilon - Y_M$$
(11)

$$\frac{\partial}{\partial x_{i}}(\rho \epsilon V_{i}) = \frac{\partial}{\partial x_{i}} \left[ \left( \mu + \frac{\mu_{t}}{\sigma_{\epsilon}} \right) \frac{\partial \epsilon}{\partial x_{i}} \right]$$

$$+C_{1\epsilon}\frac{\epsilon}{k}(G_k+C_{3\epsilon}G_b)-C_{2\epsilon}\rho\frac{\epsilon^2}{k}-R_{\epsilon}+S_{\epsilon}$$
 (12)

In Eq. (11) and (12),  $G_{\kappa}$  and  $G_b$  represent the turbulence kinetic energy generation due to the mean velocity gradients and buoyancy, respectively.  $Y_M$  is the component of the total dissipation rate attributed to the fluctuating dilatation.  $\sigma_{\kappa}$  and  $\sigma_{\varepsilon}$  denote the turbulent Prandtl numbers of k and  $\varepsilon$ , respectively.  $C_{I\varepsilon}$ ,  $C_{Z\varepsilon}$ , and  $C_{3\varepsilon}$  are model constants.

The P-1 radiation model is expressed as in Eq. (13) for radiation flux,  $q_r$ , which can be considered in the energy equation for the heat source [34].

$$-\nabla q_r = aG - 4a\sigma T^4 \tag{13}$$

In Eq. (13), G is the incident radiation,  $\sigma$  is the Stefan-Boltzmann constant, and a represents the absorption coefficient.

# 2.2 Solving Process

The combustion model was simulated for biogas fuel under stoichiometric oxy-fuel combustion at a constant power capacity of 0.5 MW supplied from the fuel. Biogases were regarded as a mixture composed of  $CH_4$  and  $CO_2$  components [14, 17, 42]. The fuels used for simulation are summarized in Table 2.

Table 2. The fuel components and lower heating values.

Definition	Components (by volume)	*LHV[MJ/kg]
M60C40	60%CH <sub>4</sub> + 40% CO <sub>2</sub>	17.6471
M70C30	$70\% CH_4 + 30\% CO_2$	22.9508
M80C20	$80\% CH_4 + 20\% CO_2$	29.6296
M90C10	90%CH <sub>4</sub> + 10% CO <sub>2</sub>	38.2979

<sup>\*</sup> The values were calculated as given in Reference [7].

Considering the fuel components and the lower heating values, the fuel mass flow rates were calculated corresponding to a power of 0.5 MW from fuel input. Thus, the oxygen mass flow rates were determined for a stoichiometric mixture for each fuel component. The pressure and temperature of the fuel and oxygen inputs are considered at atmospheric pressure and 298.15 K, respectively. Besides, the walls of the modelled furnace were subjected to adiabatic boundary conditions, meaning no heat transfer on the walls. The equations were solved with the pressure-based solver and SIMPLE algorithm scheme for

pressure-velocity coupling. The second-order upwind schemes were employed for spatial discretization of the equations. Thus, the equations were solved with the convergence criteria of  $10^{-7}$  for the energy and P1 model equations and  $10^{-3}$  for the continuity equation and others.

With the CFD model study, temperature distribution and combustion products were attained for combustion characteristics and exergy analysis for each fuel stated in Table 2.

## 2.3 Exergy Analysis

Exergy analysis is critical to an assessment of the combustion process since it provides valuable insight into irreversibility and reveals potential being able to be further utilized in a thermodynamic system or process [20]. Therefore, the present work focused on the exergy evaluation of oxy-biogas combustion for different CO<sub>2</sub> contents. Considering the combustion system in the current study, the system is a steady-flow combustion process without exergy input/output by both heat and work. Therefore, the exergy balance equation is expressed as follows:

$$E\dot{x}_r - E\dot{x}_p - E\dot{x}_d = 0 \tag{14}$$

where, r, p, and d indicate reactant, products, and destruction, respectively. The reactants in the combustion process are fuel and oxygen. Thus;

$$E\dot{x}_r = \dot{m}_f e x_f + \dot{m}_{02} e x_{02} \tag{15}$$

where,  $ex_f$  and  $ex_{O2}$  are the specific exergy (kJ/kg) of fuel and oxygen, respectively. The  $\dot{m}_f$  and  $\dot{m}_{O2}$  denote the mass flow rates of fuel and oxygen in the reactant.

Similarly, the exergy of the product mixture consisting of six species (O<sub>2</sub>, CO<sub>2</sub>, CO, H<sub>2</sub>O, H<sub>2</sub>, and CH<sub>4</sub>) in the current study is calculated by the following equation.

$$E\dot{x}_p = \dot{m}_p e x_p \tag{16}$$

In Eq. (16),  $ex_p$  represents the specific exergy of products. The  $\dot{m}_p$  is the mass flow rate of the products, which equals the sum of the mass flow rates of fuel and oxygen.

The exergy in the combustion process can be divided into two parts, thermo-mechanical exergy ( $ex_{tm}$ ) and chemical exergy ( $ex_{ch}$ ). Therefore, exergy expression can be written as follows:

$$ex = ex_{tm} + ex_{ch} (17)$$

By eliminating the contribution of kinetic and potential energy, the specific thermo-mechanical exergy for an ideal gas can be expressed as follows [43]:

$$ex_{tm} = (h - h_0) - T_o \left( c_p \ln \frac{T}{T_o} - R \ln \frac{P}{P_0} \right)$$
(18)

where, h is the specific enthalpy of the mixture of reactants or products.  $c_p$  and R are the specific heat capacity and the gas constant, respectively. The notation "o" indicates the dead state where the system is in thermo-mechanical equilibrium with its surrounding.

The chemical exergy of an ideal gas mixture is calculated as follows [44]:

$$ex_{ch} = \frac{1}{MW_{mix}} \left( \sum y_i \ ex_{ch,i}^o + R_u T_o \sum y_i ln y_i \right) \tag{19}$$

where,  $MW_{mix}$  is the molecular weight of the mixture of reactants or products.  $y_i$  is the mole fraction of species i in the mixture.  $R_{tt}$  is the universal gas constant.  $ex_{ch,i}^{o}$  is the standard molar chemical exergy of the species i. The standard molar chemical exergy is calculated by considering the mole fraction of each species at the reference environment, which is presented in Table 3 [45], as given by:

$$ex_{ch,i}^o = -RT_o \ln(y_{i,o}) \tag{20}$$

As shown in Table 2, the mole fraction of CH<sub>4</sub> is not present in the reference environment. Therefore, it was determined by utilizing the formation reaction of CH<sub>4</sub> and the standard Gibbs free energy of the species in the reaction, as in the reference [46]. Table 3 also summarizes the calculated standard molar chemical exergy.

Table 3. The reference molar fraction of the species and its standard molar exergy.

Species	Mole Fraction (y <sub>i,0</sub> )	Std. Molar ChemicalExergy $(ex_{ch,i}^{o})$ (kJ/mol)		
N <sub>2</sub> (g)	0.7567	0.6917		
$O_2(g)$	0.2035	3.9465		
$CO_2(g)$	0.000345	19.7611		
CO(g)	0.00007	29.4226		
$H_2O(g)$	0.0303	8.6675		
$H_2(g)$	0.0000005	35.9643		
CH <sub>4</sub> (g)	-	829.9431		

Thus, the exergy destruction in combustion process for each fuel was calculated using Eq. (14) as given in Eq. (21).

$$E\dot{x}_d = E\dot{x}_r - E\dot{x}_p \tag{21}$$

Finally, incomplete combustion exergy loss fraction ( $f_{E,ic}$ ) was determined as follows [47]:

$$f_{E,ic} = \frac{Ex_{ch,p}}{Ex_{ch,r}} x 100 \tag{22}$$

where, the notations "ch,p", "ch,r" denote the chemical exergy fractions of the products (exhaust gas) and reactants in the combustion process.

# 2.4 Mesh Dependency

The grid structures were created meshed with hexahedral cells by considering flow characteristics and the near-wall treatment model, which is the standard wall function in the present study. Hence, adequately fine mesh structures were created in the region of fuel and air streams in which strain flows may occur. On the other hand, course mesh structures were created on the near walls of the furnace to provide  $30 < y^+ < 300$  of which the near-wall model requires. To determine the grid effect on the result, the simulation was performed in different grid structures for oxy-methane

combustion with the JL\_modified mechanism. The effect of the grid structures is given in Table 4. Since the further increase in the cell number did not change the results considerably, the simulation in the current study was performed on the grid structure with 169605 cells.

Table 4. The effect of the grid structure on the exhaust gas parameters.

	Cell	T . DZ1	Mole Fraction				
	number	T <sub>exh</sub> [K]	CO <sub>2</sub>	CO	O <sub>2</sub>	H <sub>2</sub> O	
Mesh1	93897	1933.487	0.3195	5.795e-6	0.0412	0.6285	
Mesh2	169605	1933.399	0.3106	5.892e-6	0.0406	0.6221	
Mesh3	214263	1931.824	0.3187	5.874e-6	0.0409	0.6202	

## 3. Validation

The model was validated by comparing with the data of the temperature and velocity distribution obtained at a 0.8 MW furnace capacity with the JL\_modified reaction mechanism presented by Yin et al. [32]. As in the reference, the validated simulation was carried out using the same mass flow rates of the fuel and oxygen inputs for oxy-fuel combustion. Besides, the boundary condition on the walls was subject to a varying temperature profile with respect to axial distance. Figure 2 shows the combustion temperature and x-velocity variations at 142 cm from the burner inlet. The simulation results are compatible with those of Reference 32. Thus, the simulation model can predict reliable results for the evaluation of oxy-biogas combustion.

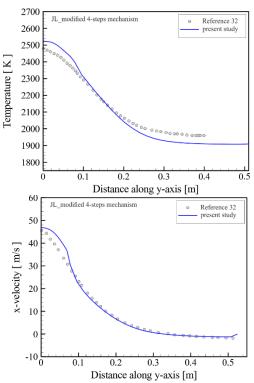


Figure 2. Comparison of the simulation results at 142 cm from the burner in the furnace with a 0.8 MW capacity.

# 4. Results

# **4.1 Temperature Development**

In Figure 3, axial temperatures are given comparatively in combustion processes using M60C40, M80C20, M70C30, and M90C10 fuel types. Accordingly, it is seen that the maximum temperature is obtained with the M90C10 fuel

type. The lowest temperature was recorded in the M60C40 fuel type. It was determined that the temperature is directly proportional to the  $CH_4$  content of the fuel and inversely proportional to the  $CO_2$  content. This can be explained by inert gases such as  $CO_2$  absorbing heat from the combustion process and acting as flame inhibitors. The presence of  $CO_2$  therefore reduces the speed of the flame burning and the combustion rate (the rate at which reactants are converted to combustion products) [48].

Similarly, Sivri et al. [49] tested biogas mixtures containing different amounts of  $CO_2$  in their study on the combustion characteristics of biogas mixtures. They found that high  $CO_2$  concentration decreased the combustion rate and caused lower axial temperature values. Ghenai and Janajreh [50] also argued that the dilution of  $CH_4$ -containing fuels with  $CO_2$  and nitrogen reduces the flame temperature. Furthermore, the decreases in temperature can also be attributed to the reduction in the calorific value of the fuel with the increasing  $CO_2$  content.

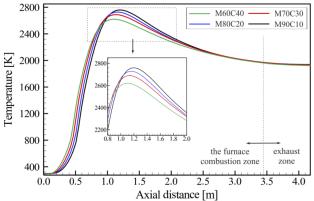


Figure 3. Comparison of the axial temperature values for different fuels.

The velocity changes along the x-axis in the combustion chamber for different fuels are given in Figure 4. Velocity profiles for the fuels show similar trends. As seen in the figure, velocity values in the combustion chamber are high at x=0.

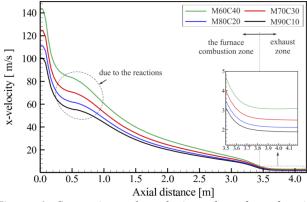


Figure 4. Comparison of x-velocity values through axial distance for different fuels.

After the fuel and oxygen entered the combustion chamber, the velocities have decreasing trends. However, the velocities partially increased between 0.5 m and 1.0 m of the axial distance due to the reaction heat. Afterward, the velocity values along the axis slowed down as the effect of the reaction heat decreased and exited the exhaust region with low velocities. Comparing the velocity values, the

velocity values in the case of M60C40 fuel are the highest among the others, and the velocity values decrease with decreasing CO<sub>2</sub> content of biogas. This is due to a decrease in mass flow rate at the same furnace capacity (see Table 5).

Figure 5 illustrates the temperature distributions along the y-axis at x=1.1 m, where the maximum axial temperature occurs approximately (see Figure 3). It is observed that the axial temperature distribution increases with the decrease of the  $CO_2$  ratio in the fuel. Therefore, the highest temperature values were obtained in the M90C10 fuel, while the lowest were in the M60C40 fuel. The reaction zones have high-temperature gradients [51]. Therefore, it can be stated that the change in temperature values after roughly y=0.3 m remains nearly constant, which corresponds to the flame radius at the x=1.1 m.

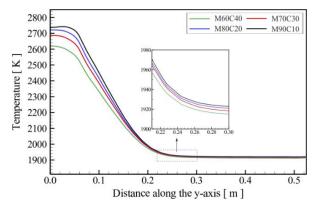


Figure 5. Comparison of temperature values with the y- axis at x=1.1 m.

Figure 6 shows the x-velocity variations in the y-axis direction at the x=1.1 m point for each fuel. As shown in the figure, the velocities are the highest in the center of the combustion chamber due to the highest flame temperature. The x-velocities decrease away from the center of the combustion chamber due to the decreasing or completing reactions, as compatible with the trends in temperature variation given in Figure 5.

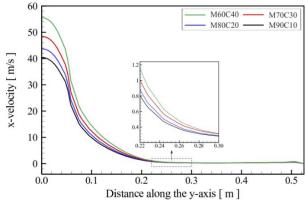


Figure 6. Comparison of x-velocity values with the y-axis at x=1.1 m.

Furthermore, the velocity gradients with the y-axis are nearly zero after x=0.3 m, which is similar to temperature variations. Generally, it is seen that the highest velocity profile belongs to M60C40 fuel. As mentioned above, the mass flow rate at the same furnace capacity gets bigger with the increasing  $CO_2$  content of biogas, which elevates the velocity values.

Figure 7 illustrates the obtained maximum temperature values in the combustion zone and the exhaust temperature variation versus CO<sub>2</sub> content in biogas. It is seen that the increasing CO<sub>2</sub> content brings about a decrease in both the maximum temperature and exhaust temperature. The increasing carbon dioxide with high heat capacity gives rise to decreasing combustion temperature, which in turn decreases the average exhaust temperature [52]. The maximum temperature values were obtained from about 2765 K to 2625 K by varying the CO<sub>2</sub> content. Besides, the exhaust temperatures varied between 1926 K and 1938 K, meaning the average temperature drop is approximately 12 K between M90C10 and M60C40 fuels. A similar trend in the drop in exhaust temperature at a constant power was reported by Bastani et al. [53].

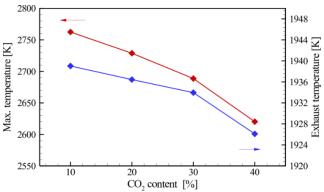


Figure 7. The variation of maximum temperature and exhaust temperature with  $CO_2$  content in biogas.

## 4.2 Exergetic Evaluation

The section is dedicated to presenting the exergy analysis results. As stated in Section 2.3, the combustion chamber exergy input is provided by reactants consisting of biogas and oxygen at atmospheric pressure and ambient temperature. Therefore, the part of its chemical exergy is only considered by neglecting thermo-mechanical exergy. The calculated specific exergy and exergy flow rates are presented in Table 5 for each fuel. As expected, the maximum specific exergy belongs to M90C10 since it has the highest concentration of  $CH_4$  which has a higher standard molar chemical exergy than that of  $CO_2$  (see Table 3).

Table 5. The specific exergy values and exergy flow rates of the reactants for the fuels.

Fuels	Specific Exergy [kJ/kg]		Mass Flow Rate [kg/s]		Exergy Flow Rate [kW]	
	$ex_f$	$ex_{o2}$	$\dot{m}_f$	$\dot{m}_{O2}$	$E\dot{x}_r$	
M90C10	39793.5680		0.0131	0.04	524.4602	
M80C20	30864.1802	123.3281	0.0169		525.7661	
M70C30	23990.7491	123.3281	0.0218		527.5887	
M60C40	18536.8392		0.0283		530.1435	

Besides, the exergy flow rates for each fuel are calculated by the sum of the exergy flow rates of the fuels and oxygen as given in Eq. (15). As shown in the table, the exergy flow rates of the reactant relatively increase by increasing  $CO_2$  content due to the increasing mass flow rate calculated by considering the same energy flow rate, 0.5 MW.

Generally, exergy values of fuels are determined by multiplying the lower heating value of fuel (LHV) and the coefficient  $\phi$  which is the ratio of the specific exergy to the lower heating value and calculated depending on the related fuel content [54]. Similar manner, the correlation developed

in the current study for biogas fuel for different  $CO_2$  contents ( $f_{co2}$ -by volume) is presented below:

$$0 < f_{co2} \le 0.4$$

$$\phi = 1.0374 + 0.056012(f_{co2})^{1.60193}$$

$$ex_{f,biogas} = \phi x LHV$$
(23)

The exergy of the exhaust gases was determined by considering the exhaust gas temperature values (given in Figure 7) and combustion products for each fuel. The obtained mass fractions of each species are given in Table 6. The specific exergy values of the exhaust gas are illustrated in Figure 8. M90C10 shows the highest specific exergy value by 6753.6 kJ/kg, consisting of 91.7 % thermo-mechanical exergy part and 8.3 % chemical exergy. On the other hand, the lowest specific exergy of the exhaust gas was attained at M60C40 by 5985.5 kJ/kg, including 82.3 % thermo-mechanical exergy and 17.7 % chemical exergy fractions.

*Table 6. Mass fraction of the combustion products.* 

Fuels	Species						
rueis	CO <sub>2</sub>	CO	H <sub>2</sub> O	$\mathbf{H}_2$	O <sub>2</sub>	CH <sub>4</sub>	
M90C10	0.5857	1.44e-5	0.4027	0.0099	0.00121	0.00051	
M80C20	0.6064	1.56e-5	0.3767	0.0153	0.00096	0.00063	
M70C30	0.6260	1.95e-5	0.3474	0.0247	0.00089	0.00091	
M60C40	0.6407	2.91e-5	0.3189	0.0379	0.00077	0.00164	

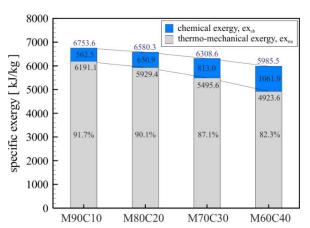


Figure 8. The specific exergy values of exhaust gas mixtures.

Figure 9 shows the exergy flow rates in the furnace. As understood in the figure, the highest exergy destruction  $(\vec{E}x_d)$  occurred in M90C10 fuel with 166.14 kW, while the lowest is in M60C40 with 121.14 kW.

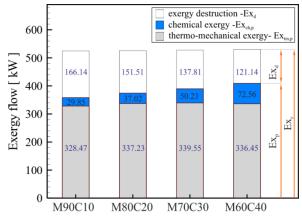


Figure 9. Exergy flow rates for biogas combustion with varying  $CO_2$  content in the furnace.

As a general result, the reduction of  $CO_2$  in biogas causes an increase in the exergy destruction flow in the current combustion system. Another significant result is that the chemical exergy flow of exhaust gas  $(Ex_{ch,p})$  rises with increasing  $CO_2$  content in biogas. Therefore, the chemical exergy flow in the exhaust reached 72.56 kW in the case of M60C40.

The chemical exergy flow in the exhaust gives an insight into the incomplete combustion exergy loss. Therefore, the chemical exergy flow fractions due to incomplete combustion ( $f_{E,ic}$ ) are shown in Figure 10 for different CO<sub>2</sub> contents. As can be seen, the increasing CO<sub>2</sub> concentration influences considerable incomplete combustion, and the  $f_{E,ic}$  value reached about 13.8 % at 40 % CO<sub>2</sub> content (M60C40) while it is about 5.8 % at 10 % CO<sub>2</sub> (M90C10).

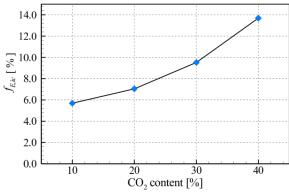


Figure 10. Exergy loss ratio due to incomplete combustion, f<sub>e.ic</sub>.

## 5. Conclusion

This study investigates the exergy analysis of oxy-biogas combustion in the furnace with a 0.5 MW input power capacity and the adiabatic condition. The focus is on how the varied biogas content with CO<sub>2</sub> ratio influences the exergy flow rates involving thermo-mechanical exergy and chemical exergy. Therefore, the specific exergy and exergy flow rates of reactants and products in the combustion process were calculated for each biogas content varying from 10% CO<sub>2</sub> to 40% CO<sub>2</sub> with a 10% increment, and the obtained results were compared to each other. Additionally, the exergy loss ratios due to incomplete combustion were evaluated considering the chemical exergy flow in the exhaust

The main findings based on the results are the following:

- The increasing CO<sub>2</sub> concentration in biogas reduced the maximum temperature values in the combustion zone due to its high heat capacity and lowering O<sub>2</sub> concentrations in reaction zones.
- Lowering calorific values with the increase in  $CO_2$  in biogas led to an increase in the mass flow rates to maintain the input power constant, which resulted in the increasing exergy flow rates of the reactants.
- With increasing CO<sub>2</sub> content, the specific exergy of the combustion products decreased while the exergy flow rates increased due to the mass flow rates.
- The chemical exergy fractions of the combustion products increased with elevating CO<sub>2</sub> content in biogas, and the chemical exergy fraction was found to be 17.7% in biogas with 40% CO<sub>2</sub> content while 8.3% at 10% CO<sub>2</sub> content.

- The exergy destruction decreased with increasing CO<sub>2</sub> content in biogas in the current combustion system
- The increasing  $CO_2$  gave rise to an increase in the exergy loss due to incomplete combustion. Thus, the exergy loss fraction due to incomplete combustion was found to be 13.8% in biogas with 40%  $CO_2$  while 5.8% at 10%  $CO_2$  content.

It should be mentioned that one possible limitation of the study is related to the EDC model constant values. These values may differ somewhat depending on the mixture properties in the combustion chamber, which may cause uncertainties in the findings. For future work, further analysis can be performed with different burner and combustion chamber designs, which are also significant factors along with fuel content. These attempts are needed to establish more efficient combustion systems for sustainability.

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