THERMO-ENVIRONMENTAL ANALYSIS AND MULTI-OBJECTIVE OPTIMIZATION OF PERFORMANCE OF ERICSSON ENGINE IMPLEMENTING AN EVOLUTIONARY ALGORITHM

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ABSTRACT

This paper makes attempt to optimize a high-temperature differential Ericsson engine with several conditions. A mathematical approach based on the finite-time thermodynamic was proposed with the purpose of gaining thermal efficiency, the output power and the entropy generation rate throughout the Ericsson system with regenerative heat loss, finite rate of heat transfer, finite regeneration process time and conductive thermal bridging loss. In this study, an irreversible Ericsson engine is analyzed thermodynamically in order to optimize its performance. In addition, three Scenarios in multi-objective optimization are presented and the results of them are assessed individually. The first strategy is proposed to maximize the Ecological function, the thermal efficiency and the Exergetic performance criteria. Furthermore, the second strategy is suggested to maximize the Ecological function, the thermal efficiency and Ecological coefficient of performance. The third strategy is proposed to maximize the Ecological function and the thermal efficiency and Dimensionless ecological based thermo-environmental function. Multi-objective evolutionary algorithms based on NSGA-II algorithm was applied to the aforementioned system for calculating the optimum values of decision variables. Decision variables considered in this paper including the regenerator's effectiveness, the high-temperature heat exchanger's effectiveness, the low-temperature heat exchanger's effectiveness, the working fluid temperature in the low-temperature isothermal process and the working fluid temperature in the high-temperature isothermal process. Moreover, Pareto optimal frontier was achieved and an ultimate optimum answer was chosen via three competent decision makers comprising LINMAP, fuzzy Bellman-Zadeh, and TOPSIS approaches. The results from scenarios shown that third scenario is the best scenario.

Keywords: Evolutionary Algorithms, Decision-Making, Thermodynamic Analysis, Multi-Objective Optimization, Entropy Generation, Ericsson Engine

INTRODUCTION

One of the simplest types of external-combustion engines is the Ericsson engine which employs a compressible fluid as a working fluid. At Carnot efficiency, the Stirling and Ericsson engines can supposedly be an effective engine to convert heat into mechanical work. Little research has been done on the Ericsson engine and more research is related to the Stirling engine.

The material employed for Stirling engine and Ericsson engine construction effects on the thermal boundary for the operation of this engine. In most cases, the engines work with a cooler and heater temperature of 338 and 923 K, correspondingly [1]. The range of efficiency in Stirling engines vary from 30 to 40% which yielded by normal operating speed varies from 2000 to 4000 rpm, and a usual temperature changes from 923 to 1073 K [2].

Several scholars propose isothermal models like Schmidt's original work (for instance, Urieli and Berchowitz [3], Reader [4] and Hargreaves [5]). Carlson and colleagues [6] improve an ideal approach with nonisothermal heat exchange. These type of approaches propose a development on the isothermal approach as they exclude unfeasibly slow engine speed accompanying with isothermal working spaces and the requirement for infinite heat transfer. Urieli and Kushnir [7] depicted that this analysis can be employed for the purpose of examining the different practical impacts of heat exchangers, non-ideal regenerators, comprising pressure losses and heat transfer. Martaj and colleagues [8] worked on the steady-state operation and illustrated a thermodynamic analysis of a low-temperature Stirling engine, and entropy, energy and exergy balances were reported at each principal component of the engine. The main aims of the Stirling engine inventors can be *This paper was recommended for publication in revised form by Regional Editor Alibakhsh Kasaeian*

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categorized into three main groups: maximum power; maximum efficiency; minimum expenses. Markman and colleagues [9] investigated the mechanical-power losses and thermal-flux of a 200W beta-configuration of the Stirling engine to improve and intensify the efficiency of engine.

A small Stirling engine with 4 watt output power in 900 rpm rotation speed and 0.1 MPa pressure was investigated and constructed by Kagawa and colleagues [10]. Brandhorst and Chapman [11] established a 5 kW engine for usage as a power generator in space usages. Ataer [12] used the Lagrangian approach to analyze the regenerators of Stirling cycle engines. Nakajima and colleagues [13] established a 10 g micro Stirling machine with an approximately 0.05 cm³ piston swept volume. At 10 Hz, the output power of the engine was 10 mW.

Aramtummaphon [14] evaluated an open cycle Stirling engines by employing steam heated from producer gas. The first engine at a maximum speed of 950 rpm produced a specified power of about 1.36 kW, whereas the second engine, amended from the first one, at a maximum speed of 2200 rpm generated a specified power of about 2.92 kW. Takashi Fukui and colleagues [15] invented and built a micro-engine, and its experimental evaluation was implemented; though, the performance of the micro-engine cannot be scaled to the real one. Iwamoto and colleagues [16] compared the efficiency of high temperature and low-temperature Stirling engines with other types of Stirling engines. They investigations depicted that the LTD Stirling efficiency is about 50% of Carnot efficiency with a similar condition. Wu and colleagues [17] depicted the impacts of regeneration time, heat transfer, and inadequate regeneration on the efficiency of the irreversible Stirling engine cycle. Erbay and Yavuz [18] studied the practical Stirling heat machine for maximum power output circumstances by employing polytropic progressions. They also specified the compression ratio and efficiency at maximum power density and determined the thermal design constraints. Ahmadi and colleagues [19] studied the effects of the Solar Collector Design variables on the Efficiency of Solar Stirling Engine. Ahmadi and colleagues [20-23] proposed a grey-box method to predict the power of Stirling heat engine via machine learning methods.

Investigation of thermodynamic irreversibility in systems acquired significance after the oil crisis in the 1970s to achieve higher efficiencies. The novel approach was entitled as Finite- Time-Thermodynamics (FTT). Primary studies in this field focused on endoreversible power cycle. This engine, named Curzon-Ahlborn-Novikov (CAN), is reversible internally and irreversible externally [24,25]. Compared with Carnot cycle, which works completely irreversible, Can engine yield more realistic results. Moreover, several studies were conducted on the maximum extractable work from irreversible systems [26–28]. For instance, Angulo-Brown established a standard known as ecological function (ECF) [29]. Yan suggested using ambient temperature (T_0) instead of heat sink temperature in Carnot efficiency [30]. Various studies are represented in the literature which has worked on ecological optimization [31–62]. ECOP is another thermo-ecological criterion which is defined and utilized in different thermodynamic cycles [63–72]. A performance coefficient, called exergetic performance criteria (EPC), is another criterion established to find out the relationship between exergy and exergy destruction of a [73–77]. Some studies [78–89] work on obtaining an approach for exergy application in finite time thermodynamic (FTT).

Several studies presented mathematical methods to calculate the overall thermal efficiency of solar powered high-temperature differential dish Stirling engine with regenerator irreversibility and finite heat transfer. Afterwards, the thermal efficiency and absorber working temperature were optimized. [90-92]. Tlili investigated the effects of regenerator effectiveness and internal irreversibility on the thermal efficiency of an endoreversible Stirling heat engine at maximum power condition [93]. Kaushik et al [94-97] studied effects of regeneration and heat transfer of the heat sink and sources on exergy destruction of Stirling and Ericsson engines. Evolutionary algorithms (EA) were originally used throughout the mid-eighties in an effort to unravel the puzzle of this general category [98]. A practical answer to a multi-objective puzzle is to determine a group of answers, each of which fulfills the objectives at a satisfactory degree without being overshadowed by any other answer[99]. Multi-objective optimization issues generally serve a feasibly innumerable group of answers that is to say as Pareto frontier, where examined vectors denote the preeminent probable trade-offs in the objective function area. In this regard, multi-objective optimization of various energy cycles was investigated by numerous nowadays [100-140]. Ahmadi and colleagues [103,105] employed NSGAII to optimize the economic and thermodynamic of a solar-dish Stirling heat engine. They presented another model to evaluate the cooling load of Stirling cryogenic refrigerator cycle as well [113]. Sayyaadi et al. [117] used NSGAII in order to optimize the design elements of a Solar-Driven Heat Engine. Ahmadi and colleagues [121] used MOEA and thermodynamic analysis to optimize an irreversible three-heat-source absorption heat pump. Sadatsakkak et al. [124,125] used thermoeconomic analysis and MOEA to optimize an irreversible regenerative closed Brayton cycle and an endoreversible Braysson cycle.

In this study, an irreversible Ericsson engine is thermodynamically investigated in order to optimize its performance. In addition, three scenarios are considered in optimization and obtained results are evaluated. The first strategy is proposed to maximize the Ecological function, the thermal efficiency and the Exergetic performance criteria. Furthermore, the second strategy is suggested to maximize the Ecological function, the thermal efficiency and ECOP. The third strategy is proposed to maximize the Ecological function and the thermal efficiency and Dimensionless ecological based thermo-environmental function. MOEAs jointed with NSGA-II approach was executed in this paper. Decision parameters involved in this paper including the regenerator's effectiveness, the high-temperature heat exchanger's effectiveness, the low-temperature heat exchanger's effectiveness, the working fluid temperature in the low-temperature isothermal process and the working fluid temperature in the high-temperature isothermal process. Moreover, Pareto optimal frontier was achieved and an ultimate optimum answer was chosen via three competent decision makers comprising the LINMAP, fuzzy Bellman-Zadeh, and TOPSIS approaches.

SYSTEM DESCRIPTION

Figure 1 depicts a graphical illustration of an Ericsson heat engine cycle with regenerative heat losses and finite-time heat transfer. As illustrated in Figure 2, ideal Ericsson cycle comprises of 4 progressions containing two isobaric progressions (2–3 and 4–1) in the regenerator and two isothermal (1–2 and 3–4). In a real cycle, it is unfeasible to have an ideal heat transfer in the regenerator, in which the complete amount of absorbed heat (in the process 4–1) is transmitted to the working fluid in the isobaric heating progression (process 2–3). Consequently, a heat transfer loss happens in the regenerator.



Figure 1. Schematic of proposed Ericsson heat engine



Figure 2. Ericsson engine T-S diagram

THERMODYNAMIC ANALYSIS OF THE SYSTEM

It is worth to stress that the finite heat transfer in the regenerative heat transfer (Q_r) can be calculated by following expression [20,91,92]

$$Q_{r} = nC_{p}\varepsilon_{R}\left(T_{h} - T_{c}\right)$$
⁽¹⁾

 ΔQ_r stands for the heat loss throughout the two regenerative progressions in the cycle and can be determined via the below equation [20, 91,92]:

$$\Delta Q_{\rm r} = nC_{\rm p} \left(1 - \varepsilon_{\rm R}\right) \left(T_{\rm h} - T_{\rm c}\right) \tag{2}$$

 C_p represents the working fluid's specific heat capacity in the regenerative progressions (mole), n stands for the mass of the working fluid in terms of mole, ε_r denotes the regenerator's effectiveness, T_h and T_c stand for the working fluid temperatures in the cold space and hot space, correspondingly.

It is not reasonable to pay no attention to the time of two regeneration progressions when compared with two constant temperature progressions included in the suggested approach. So, via the below equation the regeneration time calculated [94-97]:

$$t_{\rm R} = 2\alpha \left(T_{\rm h} - T_{\rm c} \right) \tag{3}$$

The heat absorbed between the heat sink and working fluid (Q_L) and the heat released between working fluid and heat source (Q_H) , are calculated via the below equations

$$Q_{\rm H} = nRT_{\rm h}Ln\lambda + nC_{\rm p}\left(1 - \varepsilon_{\rm R}\right)\left(T_{\rm h} - T_{\rm c}\right)$$
⁽⁴⁾

$$Q_{L} = nRT_{c}Ln\lambda + nC_{p}(1 - \varepsilon_{R})(T_{h} - T_{c})$$
(5)

where $\lambda = \frac{p_1}{p_2} = \frac{p_4}{p_3}$ is Ratio of pressure throughout the regenerative progressions. Consequently, we have:

$$Q_{\rm H} = \left[C_{\rm H} \varepsilon_{\rm H} \left(T_{\rm H} - T_{\rm h} \right) + \xi C_{\rm H} \varepsilon_{\rm H} \left(T_{\rm H}^4 - T_{\rm h}^4 \right) \right] t_{\rm h}$$
(6)

$$Q_{L} = C_{L} \varepsilon_{L} \left(T_{c} - T_{L} \right) t_{l}$$
⁽⁷⁾

in which C_L and C_H denote the external fluids heat capacitance rate in the heat sink and heat source, correspondingly.

$$\varepsilon_{\rm H} = 1 - e^{-N_{\rm H}} \tag{8}$$

$$\varepsilon_{\rm L} = 1 - e^{-N_{\rm L}} \tag{9}$$

in which $\varepsilon_{\rm H}$ and $\varepsilon_{\rm L}$ stand for the high and low temperature heat exchangers effectiveness, correspondingly and $N_{\rm L} = {}^{U_{\rm L}A_{\rm L}} / C_{\rm L}$, $N_{\rm H} = {}^{U_{\rm H}A_{\rm H}} / C_{\rm H}$ represent the cyclic period. Using Eqs. (3)- (9), we get that the cyclic period *t* is:

$$t = \frac{nRT_{h}Ln\lambda + nC_{p}(1-\varepsilon_{R})(T_{h}-T_{c})}{C_{H}\varepsilon_{H}(T_{H}-T_{h}) + \xi C_{H}\varepsilon_{H}(T_{H}^{4}-T_{h}^{4})} + \frac{nRT_{c}Ln\lambda + nC_{p}(1-\varepsilon_{R})(T_{h}-T_{c})}{C_{L}\varepsilon_{L}(T_{c}-T_{L})} + 2\alpha(T_{h}-T_{c})$$
(10)

Take into account the cyclic period of the Ericsson engine, the thermal efficiency, the output power, and entropy production of the engine can be determined as following as:

$$P = \frac{W}{t} = \frac{Q_H - Q_L}{t}$$
(11)

$$\eta_t = \frac{Q_H - Q_L}{Q_H} \tag{12}$$

$$\sigma = \frac{1}{t} \left(\frac{Q_L}{T_L} - \frac{Q_H}{T_H} \right)$$
(13)

Exergy destruction is the measurement of the irreversibilities or lost work in the system and it is equal to environment temperature (T_0 , K) multiply entropy generation rate (kW/K). The rate of Exergy destruction (kW) is written as following:

$$\mathbf{E}_{\mathrm{xD}} = \mathbf{T}_0 \boldsymbol{\sigma} \tag{14}$$

Exergy efficiency:

$$\varphi = \frac{P}{P_R}$$
(15)

Ecological function (kW) can be defined as:

$$\mathbf{E} = \mathbf{P} - \mathbf{E}_{\mathbf{x}\mathbf{D}} \tag{16}$$

The ECOP and exergetic performance criteria are calculated as following as:

$$ECOP = \frac{P}{T_0 \sigma}$$
⁽¹⁷⁾

$$EPC = \frac{Ex}{T_0 \sigma}$$
⁽¹⁸⁾

Reversible work per unit time of the system (kW) is the difference of exergy input of the system and exergy output from the system and it is described as following:

$$P_{\rm R} = \frac{1}{t_{\rm cycl}} \left(Q_{\rm H} \left(1 - \frac{T_0}{T_{\rm H}} \right) - Q_{\rm L} \left(1 - \frac{T_0}{T_{\rm L}} \right) \right)$$
(19)

where m is the total mass, b_1 is environmental impact of the components (mPts/kg), b_2 is the environmental impact of the fuel (mPts/ MJ), b_3 is the environmental impact of the lost work (mPts/MJ), and b_4 (mPts/MJ), is the environmental impact of the mechanical work. Similarly, the ecological based thermo-environmental function is defined as:

$$B_{E} = \frac{E}{b_{1}m + b_{2}Q_{H} + b_{3}E_{xD} + b_{4}W}$$
(20)

The dimensionless ecological based thermo-environmental function is defined as:

$$b_E = b_4 B_E \tag{21}$$

Substituting Eqs. (3)- (10) into Eqs. (11) and (12) we have,

$$P = \frac{nR(T_h - T_c)Ln\lambda}{\frac{nRT_hLn\lambda + nC_p(1 - \varepsilon_R)(T_h - T_c)}{C_H\varepsilon_H(T_H - T_h) + \xi C_H\varepsilon_H(T_H^4 - T_h^4)} + \frac{nRT_cLn\lambda + nC_p(1 - \varepsilon_R)(T_h - T_c)}{C_L\varepsilon_L(T_c - T_L)} + 2\alpha(T_h - T_c)}$$
(22)

$$\eta_{t} = \frac{nR(T_{h}-T_{c})Ln\lambda}{nRT_{h}Ln\lambda+nC_{p}(1-\varepsilon_{R})(T_{h}-T_{c})}$$
(23)

MULTI-OBJECTIVE OPTIMIZATION WITH EVOLUTIONARY ALGORITHMS OPTIMIZATION VIA EA

Genetic Algorithms were firstly proposed by Prof. Holland (1960) by inspiring the concept of natural evolution and Darwinian theorem for optimization purposes [101]. The evolution typically commences from a population of accidentally created individuals and takes place in creations. In each creation, the fitness value of each individual in the population is examined; multiple individuals are stochastically chosen from the present population, and improved to create a fresh population. The fresh population is then employed in the following iteration of the GA. Usually, the GA stops when either an acceptable fitness level was achieved for the population or a maximum number of generations were created. More details of GA can be found in previous works [99, 102].

Also, MOEAs were evolved throughout the past years by frequent examinations on multipart mathematical puzzles and on practical engineering issues and have depicted that they can exclude the complications of conventional approaches [99, 102]. The construction of the MOEA employed in this paper is depicted through Figure 3 [101]. It is worth to highlight that the real values of decision parameters were employed rather than their binary codes.



Figure 3. Algorithm steps applied in the study

NSGA-II APPROACH

NSGA-II approach was employed in this paper with the purpose of determining the Pareto frontier by running GA. In this regard, NSGA-II organized the answers based on the Pareto theory and arranging non-dominated answers into non-dominated layers as illustrated in Figure 4. Put it in another way, if N_p stands for the population number, it is classified into N_L layers in which juncture of each two random chosen layers is blank assortment and combination of all layers represents N_p assortment.



Figure 4. NSGA-II solution layering

The virtual fitness of each answer is equivalent to its layer. Tournament selection was employed for cross over operating in parent choosing between two random chosen layers. So, answer placed on the layer 1, have more opportunity to be chosen for the next creation. Uniform distribution of answers along layers is regulated via an index called "index of crowding distance" for each answer. This criterion is defined as a ratio of detraction of objective functions for two neighbor answers nearby the present answer to the detraction of the minimum and maximum values of that objective. Consequently, for k_{th} objective of j_{th} answer, following expression can be used.

$$i_{dis,j,k} = \frac{f_{k,j-1} - f_{k,j+1}}{f_{k,\max} - f_{k,\min}}$$
(24)

For margin answers are allocated an infinite distance index. The summation of individual distance values conforming to each objective stands for the overall crowding distance value as follows:

$$I_{dis,j} = \sum_{k=1}^{M} i_{dis,j,k}$$
(25)

in which j represents the individual index and M stands for the number of objectives. Figure 5 depicts a graphical illustration of examination of distance index. In this approach, two variables are determined for each answer:

1) Dominant (Layer) number, N_L , namely the number of answers which control the present answer. Description and definitions of domination were described well in Ref. [98, 100]. Dominant number, for nondominated answers of the present population is equal to 0, consequently, these answers are located in layer 1. Non-dominated answers for an assortment of the answers not including the layer 1 members are located in layer 2. For M objectives issue with N populations, the number of assessments is equal to MN^2 . This process persisted with the purpose of accommodating all answers in their suitable layers. Furthermore, i rank index for each answer is allocated as its layer number, N_L . 2) Crowded comparison operator, $n\prec$, defined as following as:

A p B if
$$(\operatorname{rank}_{A} < \operatorname{rank}_{B})$$

Or:
 $((\operatorname{rank}_{A} = \operatorname{rank}_{B}) \text{ and } \operatorname{I}_{dis,A} > \operatorname{I}_{dis,B})$ (26)

It reveals that for two answers with dissimilar layers, the answer with the lower layer is desired. Else, for two answers of the same layer, the answer located in the area with a lower concentration of answers is chosen.



Figure 5. Schematic of distance indexing of individuals in NSGA-II algorithm. Distance indexing of components in NSGA-II algorithms

OBJECTIVE FUNCTION, RESTRAINTS AND DECISION PARAMETERS

The thermal efficiency, the Ecological function and the Exergetic performance criteria three objective functions for the first scenario, which are evaluated via Eqs. (12 and 16 and 18).

The Ecological function, the thermal efficiency, and ECOP three objective functions for the second scenario, which are evaluated via Eqs. (12 and 16 and 17).

The Ecological function, the thermal efficiency, and Dimensionless ecological based thermoenvironmental function three objective functions for the third scenario, which are evaluated via Eqs. (12 and 16 and 21). Throughout all scenarios proposed in the present paper the below design parameters were employed:

 ε_{R} : Regenerator's effectiveness

 ε_{L} : The low-temperature heat exchanger effectiveness

 \mathcal{E}_{H} : The high-temperature heat exchanger effectiveness

 $T_{\rm h}$: The working fluid temperature in the high-temperature isothermal process 3-4

 T_c : The working fluid temperature in the low-temperature isothermal process 1-2

Following limitations were included in the optimization process:

$$0.7 \le \varepsilon_{\rm R} \le 0.95 \tag{27}$$

$$0.7 \le \varepsilon_{\rm H} \le 0.9 \tag{28}$$

$$0.7 \le \varepsilon_{\rm L} \le 0.9 \tag{29}$$

$$800 \le T_h \le 1100 K$$
 (30)

$$350 \le T_c \le 500 K \tag{31}$$

To determine the optimal design variables of the system, based on genetic algorithm approach a simulation program was coded through Matlab software. Specifications of GA for optimization puzzle are reported in Table 1.

GA Parameters	Value
Population size	400
Population type	Double vector
Tournament size	2
Selection process	Tournament
Maximum number of generations	1000
Mutation	Restriction dependent

Table 1: Specification of GA for optimization puzzle in this paper

Choosing a final optimum answer from Pareto optimal frontier in multi-objective optimization process plays a significant role. In this regard, we should employ decision makers to determine this. Consequently, in this paper three competent decision makers including TOPSIS, Fuzzy and LINMAP were employed as decision makers. Details of these decision makers can be found in previous literature especially references [141, 142].

RESULTS AND DISCUSSION

Results of First Scenario

Via running multi-objective optimization approach the thermal efficiency, the Ecological function and the Exergetic performance criteria are maximized concurrently. The objective functions in the applied optimization, and the restrictions that were employed, are formulated by Eqs. (12 and 16 and 18) and Eqs. (27-31), respectively.

Design variables in optimization process are the low-temperature heat exchanger's effectiveness, regenerator's effectiveness, the working fluid temperature in the high-temperature isothermal process 3-4, the high-temperature heat exchanger's effectiveness, the working fluid temperature in the low-temperature isothermal process 1-2. Following specifications have been considered for Ericson cycle [91,94]:

$$n = 1, C_{v} = 15J.mol^{-1}.K^{-1}, R = 4.3J.mol^{-1}.K^{-1}, T_{H} = 1300K, T_{L} = 300K, T_{0} = 285K,$$

 $C_{L} = C_{H} = 1000 \text{ WK}^{-1}, x = 2' 10^{-10}, 1 = 2, a = 10^{-5} \text{ s/K}.$

Pareto optimal frontier for three objective functions, the objective function associated with the thermal efficiency, the Ecological function and the Exergetic performance criteria of the irreversible Ericsson engine are represented in Figure 6.

$$d_{+} = \sqrt{(\eta_{t} - \eta_{t,n})^{2} + (ECF - ECF_{n})^{2} + (EPC - EPC_{n})^{2}}$$
(32)

$$d_{-} = \sqrt{(\eta_t - \eta_{t,n,\text{non-ideal}})^2 + (\text{ECF} - \text{ECF}_{n,\text{non-ideal}})^2 + (\text{EPC} - \text{EPC}_{n,\text{non-ideal}})^2}$$
(33)

$$d = \frac{d_+}{(d_+) + (d_-)}$$
(34)

 $\eta_{t,n}$, ECF_n and EPC_n denote Euclidian the thermal efficiency, the Ecological function and the Exergetic performance criteria. Furthermore, Table 2 comprises the deviation index (*d*) for the outcome of each decision maker.



Figure 6. Pareto optimal frontier in the objectives' space of first scenario

Table 2 depicts the optimal outputs achieved for objective functions and decision parameters by executing LINMAP, Fuzzy and TOPSIS approaches for the first scenario. To determine deviations of the results from an ideal and non-ideal solution, following equations were employed.

As reported through Table 2, the deviation indexes for TOPSIS, LINMAP and Fuzzy are 0.036, 0.031 and 0.026, respectively. As clear be seen from this Table, it can conclude that the FUZZY decision-maker has a lower deviation index; consequently the answer which was chosen via the FUZZY decision-maker was selected as a final optimal answer of the multi-objective optimization for the irreversible Ericsson cycle.

Decision	Decision variables						Objectives				
Making Method	e _R	e_{H}	e_L	T_h		h_{t}	ECF(W)	EPC			
TOPSIS	0.950	0.900	0.900	1094.415	364.700	0.548	20104.500	3.721	0.036		
LINMAP	0.950	0.900	0.900	1094.390	366.479	0.547	20213.269	3.703	0.031		
Fuzzy	0.950	0.900	0.900	1094.386	368.346	0.546	20316.560	3.684	0.026		
Ideal point	-	-	-	-	-	0.554	20849.81	3.822	0		
Non-ideal point	-	-	-	-	-	0.535	19383.34	3.504	œ		

Table 2. Outcomes of the decision makers for the first scenario

To examine the accuracy of the decision maker's analysis of error was performed. Table 3 demonstrates MAPE (Mean Absolute Percentage Error) and MAAE (Maximum Absolute Percentage Error) for results achieved by the decision makers.

Decision Making Method		TOPSIS		LINMAP			Fuzzy		
Objectives	h_{t}	ECF	EPC	h_t	ECF	EPC	h_t	ECF	EPC
Max Error %	2.148	8.404	3.811	1.720	7.400	3.147	0.496	4.849	1.332
Average Error %	1.041	4.670	1.865	0.836	4.277	1.545	0.259	2.467	0.674

Table 3. Analysis of Error for the results of the first scenario

Results of Second Scenario

Three objective functions are considered for optimization which contain the thermal efficiency, the Ecological function and ECOP (should be maximized) which formulated via Eqs. (12 and 16 and 17), correspondingly.

Objective functions in this scenario are expressed by Eqs. (12 and 16 and 17) and design variables are formulated with Eqs. (27)-(31).

Design variables in optimization process are the same as the first scenario. Following specifications have been considered for Ericson cycle [91,94]:

$$n = 1, C_{v} = 15J.mol^{-1}.K^{-1}, R = 4.3J.mol^{-1}.K^{-1}, T_{H} = 1300K, T_{L} = 300K, T_{0} = 285K,$$
$$C_{L} = C_{H} = 1000 \quad WK^{-1}, x = 2' 10^{-10}, 1 = 2, a = 10^{-5} s/K.$$

Figure 7 depicts the Pareto frontier in the suggested objectives' space achieved in the optimization scenario. Three ultimate answers were chosen by the LINMAP, Fuzzy Bellman-Zadeh, and TOPSIS decision makers which are highlighted in this figure. According to Figure 7, the obtained points by LINMAP and TOPSIS are approached towards each other. Also, it was shown that the optimal value of the thermal efficiency varied from 53.467% to 54.844% and the optimal value of the Ecological function was between 19.972 (kW) and 20.850 (kW) and the optimal value of the ECOP was between 2.400 and 2.615.



Figure 7. Pareto optimal frontier in the objectives' space of second scenario

Table 4 reports the optimal outputs achieved for objective functions and decision parameters via running TOPSIS, Fuzzy and LINMAP approaches for second scenario. To determine deviations of the results from an ideal and non-ideal solution, following equations were employed.

$$d_{+} = \sqrt{(\eta_{t} - \eta_{t,n})^{2} + (ECF - ECF_{n})^{2} + (ECOP - ECOP_{n})^{2}}$$
(35)

$$d_{-} = \sqrt{\left(\eta_{t} - \eta_{t,n,\text{non-ideal}}\right)^{2} + \left(\text{ECF} - \text{ECF}_{n,\text{non-ideal}}\right)^{2} + \left(\text{ECOP} - \text{ECOP}_{n,\text{non-ideal}}\right)^{2}}$$
(36)

$$d = \frac{d_+}{(d_+) + (d_-)} \tag{37}$$

 $\eta_{t,n}$, ECF_n and ECOP_n denote Euclidian the thermal efficiency, the Ecological function and the

ECOP. Furthermore, Table 4 comprises the deviation index (d) for the outcome of each decision maker.

As reported in Table 4, the deviation indexes for TOPSIS, LINMAP and Fuzzy are 0.035, 0.030 and 0.017, respectively. As clear be seen from this Table, it can conclude that the FUZZY decision-maker has a lower deviation index; consequently the answer which was chosen via the FUZZY decision-maker was selected as a final optimal answer of the multi-objective optimization for the irreversible Ericsson cycle.

		D	ecision v	ariables		Deviation			
Decision Making Method	e_{R}	e_{H}	e_L	T_h	T_{c}	h_{t}	ECF(W)	ECOP	index
TOPSIS	0.950	0.900	0.900	1088.063	364.556	0.547	20120.226	2.594	0.035
LINMAP	0.950	0.900	0.900	1088.062	366.269	0.546	20226.707	2.576	0.030
Fuzzy	0.950	0.900	0.900	1088.039	371.284	0.543	20487.154	2.526	0.017
Ideal point	-	-	-	-	-	0.554	20849.810	2.713	0
Non-ideal point	-	-	-	-	-	0.535	19383.340	2.399	∞

Table 4. Outcomes of the decision makers for the second scenario

Finally, deviations of the final answers gained by each decision maker and ideal answer are assessed. Table 5 demonstrates MAPE and MAAE of results obtained via the aforesaid decision makers.

Decision Making Method		TOPS	IS		LINM	AP	Fuzzy		
Objectives	h_t	ECF	ECOP	h_t	ECF	ECOP	h_t	ECF	ECOP
Max Error %	2.148	8.404	3.811	1.720	7.400	3.147	0.496	4.849	1.332
Average Error %	1.041	4.670	1.865	0.836	4.277	1.545	0.259	2.467	0.674

Table 5. Analysis of error for the results of the second scenario

Results of Third Scenario

Throughout this scenario we attempted to maximize the thermal efficiency, the Ecological function and Dimensionless ecological based thermo-environmental function at the same time. The objective functions, and the limitations which were employed, are expressed by Eqs. (12 and 16 and 21) and Eqs. (27-31), respectively. Design variables in optimization process are the same as the first scenario. Following specifications have been considered for Ericson cycle [91, 94]:

$$n = 1, C_v = 15J.mol^{-1}.K^{-1}, R = 4.3J.mol^{-1}.K^{-1}, T_H = 1300K, T_L = 300K, T_0 = 285K,$$

 $C_L = C_H = 1000 \text{ WK}^{-1}, x = 2' 10^{-10}, 1 = 2, a = 10^{-5} \text{ s/K}.$

Pareto optimal frontier for three objective functions, the objective function associated with the Ecological function, the thermal efficiency, and Dimensionless ecological based thermo-environmental function of the irreversible Ericsson engine are represented in Figure 8.



Figure 8. Pareto optimal frontier in the objectives' space of third scenario

Table 6 reports the optimal outputs achieved for objective functions and decision parameters via running TOPSIS, Fuzzy and LINMAP approaches for third scenario. To determine deviations of the results from an ideal and non-ideal solution, following equations were employed.

$$d_{+} = \sqrt{(\eta_{t} - \eta_{t,n})^{2} + (ECF - ECF_{n})^{2} + (b_{E} - b_{E,n})^{2}}$$
(38)

$$d_{-} = \sqrt{(\eta_{t} - \eta_{t,n,non-ideal})^{2} + (ECF - ECF_{n,non-ideal})^{2} + (b_{E} - b_{E,n,non-ideal})^{2}}$$
(39)

$$d = \frac{d_+}{(d_+) + (d_-)} \tag{40}$$

 $\eta_{t,n}$, ECF_n and $b_{E,n}$ denote Euclidian the thermal efficiency, the Ecological function and the Dimensionless ecological based thermo-environmental function. Furthermore, Table 6 comprises the deviation index (d) for the outcome of each decision maker.

As reported through Table 6, the deviation indexes for TOPSIS, LINMAP and Fuzzy are 0.052, 0.043 and 0.035, respectively. As clear be seen from this Table, it can conclude that the FUZZY decision-maker has a lower deviation index; consequently the answer which was chosen via the FUZZY decision-maker was selected as a final optimal answer of the multi-objective optimization for the irreversible Ericsson cycle.

Decision		D	ecision v	ariables					
Making Method	e_{R}	e_{H}	e _L	T_{h}	T_{c}	h_{t}	ECF(W)	$b_{E}' 10^{3}$	Deviation index
TOPSIS	0.950	0.900	0.900	1096.679	360.197	0.552	19772.681	1.378	0.052
LINMAP	0.950	0.900	0.900	1096.679	362.556	0.550	19948.829	1.366	0.043
Fuzzy	0.950	0.900	0.900	1096.668	365.063	0.549	20115.565	1.353	0.035
Ideal point	-	-	-	-	-	0.554	20849.810	1.400	0
Non-ideal point	-	-	-	-	-	0.535	19383.340	1.200	∞

Table 6. Outcomes of the decision makers for the third scenario

Finally, deviations of the final answers gained by each decision maker and ideal answer are assessed. Table 7 explicates MAAE and MAPE of results obtained via the aforesaid decision makers.

Decision Making Method		TOPSIS			LINMAP		Fuzzy			
Objectives	h_{t}	ECF	$b_{\scriptscriptstyle E}$	h_t	ECF	$b_{\scriptscriptstyle E}$	h_t	ECF	$b_{_E}$	
Max Error %	2.148	8.404	3.811	1.720	7.400	3.147	0.496	4.849	1.332	
Average Error %	1.041	4.670	1.865	0.836	4.277	1.545	0.259	2.467	0.674	

Table 7. Analysis of error analysis for the results of the third scenario

Figure 9 depicts the comparison between thermal efficiency gained from three scenarios. As clear be seen from Figure 9, the third scenario has the highest value of thermal efficiency compared to other scenarios. Moreover, the lowest value of thermal efficiency is for the second scenario. It should be noted that thermal efficiency was an objective function for all the scenarios.



Figure 9. Comparison of thermal efficiency between different optimization scenarios

Figure 10 depicts the comparison between ECF gained from three scenarios. As clear be seen from Figure 10, the second scenario has the highest value of ECF compared to other scenarios. Moreover, the lowest value of ECF is for the third scenario. It should be noted that ECF was an objective function for all the scenarios.



Figure 10. Comparison of ECF between different optimization scenarios

Figure 11 depicts the comparison between EPC gained from three scenarios. As clear be seen from Figure 11, the third scenario has the highest value of EPC compared to other scenarios. Moreover, the lowest value of EPC is for the second scenario. It should be noted that EPC was an objective function just for the first scenario and the values of EPC in other scenarios were calculated at optimum conditions gained from optimization process.



Figure 11. Comparison of EPC between different optimization scenarios

Figure 12 depicts the comparison between ECOP gained from three scenarios. As clear be seen from Figure 12, the third scenario has the highest value of ECOP compared to other scenarios. Moreover, the lowest value of ECOP is for the second scenario. It should be noted that ECOP was an objective function just for the second scenario and the values of ECOP in other scenarios were calculated at optimum conditions gained from optimization process.



Figure 12. Comparison of ECOP between different optimization scenarios

Figure 13 depicts the comparison between Dimensionless ecological based thermo-environmental function gained from three scenarios. As clear be seen from Figure 13, the third scenario has the highest value of Dimensionless ecological based thermo-environmental function compared to other scenarios. Moreover, the lowest value of dimensionless ecological based thermo-environmental function is for the second scenario. It should be noted that dimensionless ecological based thermo-environmental function was an objective function

just for the third scenario and the values of dimensionless ecological based thermo-environmental function in other scenarios were calculated at optimum conditions gained from optimization process.



Figure13. Comparison of dimensionless ecological based thermo-environmental function between different optimization scenarios

Finally, as it is clearly seen from Figures 9 through 13, it can conclude that the third scenario was the best scenario in comparison other proposed scenarios. This is main due to the condition when b_E is maximum the values of ECOP, EPC and thermal efficiency are maximum. In other words, maximizing b_E results in gaining maximum values of ECOP, EPC and thermal efficiency.

CONCLUSIONS

This paper made attempt to illustrate multi-objective optimization of Ericson system based on finitetime thermodynamics analysis. In this regard, the optimum values of the Ecological function, the thermal efficiency, Exergetic performance criteria, the ECOP, and dimensionless ecological based thermo-environmental function of the Ericsson engine have been determined. The thermal efficiency, output power, and entropy generation rate throughout the engine have been chosen as parallel objective functions in the optimization process. Furthermore, the low temperature heat exchanger's effectiveness (\mathcal{E}_L),the regenerator's effectiveness (\mathcal{E}_R), the working fluid temperature in the high temperature isothermal process(T_h), the high temperature heat exchanger's effectiveness (\mathcal{E}_H), and working fluid temperature in the low temperature isothermal process. MOEA based on NSGA-II approach was applied to the aforementioned system for calculating the optimum values of decision variables. Moreover, Pareto optimal frontier was achieved and an ultimate optimum answer was chosen via three competent decision makers comprising the LINMAP, fuzzy Bellman-Zadeh, and TOPSIS approaches. If the main goal is ECF, the results of the second scenario are the best. Also, If the main goals are thermal efficiency, ECOP, EPC and Dimensionless ecological based thermo-environmental function the results of the third scenario are the best.

NOMECLATURE

А	Area,[m ²]
С	Heat capacitance rate,[W/K]
C _p	Specific heat capacity,[Jmol- ¹ K ⁻¹]
h	heat transfer coefficient[Wm ⁻² K ⁻¹]
n	Number of mole[-]
Ν	Number of heat transfer units[-],
Р	Power output ,[W]

Q	Heat, [J]
R	The gas constant,[Jmol ⁻¹ K ⁻¹]
S	Entropy,[J/K]
Т	Temperature,[K]
t	Time,[s]
Р	Power,[W]
р	Pressure of the working fluid
W	Output work, [J]
0	ambient
1	Inlet
1,2,3,4	State points
2	Outlet
С	Cold side
dis	distance
Н	Heat source
h	Hot side
L	Cold side/Heat sink
R	Regenerator
rec	absorber
η	Thermal efficiency[-]
ε	Effectiveness and emissivity factor[-]
λ	Ratio of pressure during the regenerative processes [-]
δ	Stefan's constant [Wm ⁻² K ⁻⁴]
σ	Entropy production [W/K]

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